

# Pearson Correlation and Discrete Wavelet Transform for Crack Identification in Steel Beams

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**Abstract:** Discrete wavelet transform is the useful means for crack identification of beam structures. However, its accuracy is severely dependent on the selecting mother wavelet and vanishing moments, which raises a significant challenge in practical structural crack identification. In this paper, a novel approach is introduced for structural health monitoring of beams to fix this challenge. The approach is based on the combination of statistical characteristics of vibrational mode shapes of the beam structures and their discrete wavelet transforms. First, this paper suggests using regression statistics between intact and damaged modes to monitor the health of beam structures. Then, it suggests extracting quasi-Pearson-based mode shape index of the beam structures to use them as an original signal in discrete wavelet transforms. Findings show that the proposed approach has several advantages compared with the conventional mode shape signal processing by the discrete wavelet transforms and significantly improves damage detection's accuracy.

**Keywords:** Pearson correlation-based damage detection; beam structures; discrete wavelet transforms; structural health monitoring

**MSC:** 37M10

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## 1. Introduction

Damage detection of structures at the early stages plays a vital role in their health [1–3]. In addition, timely damage detection saves many repair costs. Many damage detection methods have been developed to monitor structures' health and detect their damages in recent decades [4–6]. Vibration-based damage detection is a large class of damage detection methods [7–9]. The modal characteristics such as structures' stiffness, natural frequencies, and mode shapes are used to detect damage in structures [10].

In [6], it was proved that the local stiffness at the damage region reduces due to the damage. Moreover, in a study, it was reported that the existence of damage decreases the natural frequencies of the structures [7]. In practice, obtaining the structures' stiffness and natural frequencies corresponding to higher modes is impossible. Thus, mode shapes are suitable modal characteristics for simple damage detection in many structures. So far, many mode-shape-based damage detection methods have been developed by various researchers. Yazdanpanah et al. [11] proposed an indicator for damage detection in beam structures using mode shape data. Moreover, Ratcliffe et al. [12] applied a modified Laplacian indicator on the mode shapes' data to detect damages in beam structures. Dahak

et al. [13] present a damage detection approach by combining the mode shape and the curvature of beam structures. This investigation was based on the global first derivative of the beam's mode shape. However, a practical signal processing-based damage detection method called wavelet transform can process mode shapes locally. So far, many investigations have been conducted for damage detection in beam structures by using wavelet transform. Generally, there are two main classes of wavelet transforms for the beam structures' one-dimensional signal processing: one-dimensional continuous wavelet transforms (1D-CWTs) and one-dimensional discrete wavelet transforms (1D-DWTs). In these transforms, signals are processed to localize the damage's position and severity via the signal information obtained from the damaged structure [14]. Wavelets transform a signal into sub-signals to indicate the damaged signal's discontinuity [15]. Janeliukstis et al. [16] performed an empirical investigation by 1D-CWTs for damage localization of beam structure based on mode shapes' signal. Moreover, Montanari et al. [17] utilized the 1D-CWTs to localize the cracks of beam structures. Rucka et al. [18] studied the application of 1D-CWTs to detect beams' damage.

However, a significant weakness of the wavelet transform is that its accuracy depends on the amount of the vanishing moments and the mother wavelet function. In other words, changing the amount of vanishing moments may change the outcome of the damage detection. Dependence of accuracy of damage detection on the mentioned parameters is a significant disadvantage for detecting the damage because misidentifying it may have irreparable outcomes. Investigations have shown that, especially for low-level damages, this dependence is significantly increased, and the performance of the wavelet transform is impaired. In order to fix this problem and improve the accuracy of damage detection by 1D-DWTs in beam structures, this paper proposes creating a regression-based signal obtained from the correlations of the intact and damaged mode shapes. The importance of the study is that in this study, for the first time, it was suggested to use the Pearson-based coefficient of damage mode shape for feeding in 1D-wavelet transform to have accurate and robust damage detection.

## 2. Basic Formulations

### 2.1. Finite Element Modeling

The equation of motion based on the finite element model (FEM) for the free vibration on the beam is presented as follows [7]:

$$[M]\{\ddot{x}(t)\} + [K]\{x(t)\} = 0 \quad (1)$$

where  $[M]$  and  $[K]$  denote the mass and stiffness matrices in the global coordinate, respectively. Likewise, the  $\{\ddot{x}(t)\}$  and  $\{x(t)\}$  show the acceleration and displacement vectors, respectively.

By assuming the response  $\{x\} = \{\varphi\}e^{i\omega t}$ , the Equation (1) is rewritten as follows:

$$[M]\{-\omega^2\{\varphi\}\} + [K]\{\varphi\} = 0 \quad (2)$$

For the  $i^{\text{th}}$  mode, Equation (2) is rewritten as follows [19]:

$$[M]\{-\omega_i^2\{\varphi_i\}\} + [K]\{\varphi_i\} = 0 \quad (3)$$

where  $\omega_i$  shows the  $i^{\text{th}}$  natural frequency, and  $\varphi_i$  is its corresponding  $i^{\text{th}}$  mode shape. Equation (4) is named as the  $i^{\text{th}}$  eigenvalue equation. Assuming  $\omega_i^2 = \lambda_i$ , the Equation (3) can be written in the following standard form [20]:

$$[M]\{-\lambda_i\{\varphi_i\}\} + [K]\{\varphi_i\} = 0 \quad (4)$$

The matrices  $[K]$  and  $[M]$  are obtained by assembling the element stiffness matrix  $[K^e]$  and the element mass matrix  $[M^e]$ , respectively. The element stiffness matrix and the element mass matrix for the steel beam structures are expressed as follows:

$$[K^e] = \frac{EI}{l_e^3} \begin{bmatrix} 12 & 6l_e & -12 & 6l_e \\ & 4l_e^2 & -6l_e & 4l_e^2 \\ & & 12 & -6l_e \\ \text{Sym} & & & 4l_e^2 \end{bmatrix} \quad (5)$$

$$[M^e] = \frac{ml_e}{420} \begin{bmatrix} 156 & 22l_e & 54 & -13l_e \\ & 4l_e^2 & 13l_e & -3l_e^2 \\ & & 156 & 22l_e \\ \text{Sym} & & & 4l_e^2 \end{bmatrix} \quad (6)$$

The natural frequencies are obtained from the following equation:

$$\text{determinant}([K] - \lambda_i[M]) = 0 \quad (7)$$

Finally, the mode shapes are obtained from the following expression:

$$[[M]\{-\lambda_i\} + [K]]\{\varphi_i\} = 0 \quad (8)$$

## 2.2. Regression

Regression is an index which by the relationship between one or more independent variables and a dependent variable is estimated. The regression analysis has different types, and one of the most widely used regression analyses is a linear regression analysis. In this paper, linear regression examines the modal shape's data points to examine the beam structure's health.

Consider a dataset as  $\{\varphi_i^{\text{intact}}, \varphi_i^{\text{operational}}\}_{i=1}^n$  containing  $n$  data points. When  $\varphi_i^{\text{intact}}$  and  $\varphi_i^{\text{operational}}$  are considered as dependent (intact mode shape) and (damaged mode shape) independent variables; respectively, a linear regression can be expressed as follows:

$$\varphi_i^{\text{intact}} = \beta_0 + \beta_1 \varphi_i^{\text{operational}} + \epsilon \quad (9)$$

The above equation shows a line equation with an additional term called error  $\epsilon$ . The parameters of this linear model are the vertical intercept  $\beta_0$  and the slope of the line  $\beta_1$ . The slope of the line in the linear regression model indicates the sensitivity of the independent variable. The vertical intercept represents the value of the dependent variable, which is calculated as zero for the value of the independent variable. Alternatively, the constant value or the vertical intercept can be considered the average value of the dependent variable for deleting the independent variable. There are several ways to define and minimize an error  $\epsilon$ . The criterion used in the simple linear regression model is to minimize the sum of squares of error. Since the mean of the error values is zero, it is known that the sum of the squares of the error will be minimal when the data distribution is normal. As a result, the normality of the dependent variable or residual data is one of the essential assumptions for a simple linear regression model.

In order to compute the Pearson-based correlation between the intact and damaged modes, the following expression is used by setting  $\varphi_i^{\text{operational}} = \kappa_i$  and  $\varphi_i^{\text{intact}} = \zeta_i$ :

$$C(i) = \frac{(\kappa_i - \bar{\kappa}_i)(\zeta_i - \bar{\zeta}_i)}{\sqrt{(\kappa_i - \bar{\kappa}_i)^2(\zeta_i - \bar{\zeta}_i)^2}} \quad i = 1, 2, \dots, n \quad (10)$$

where  $\bar{\zeta}_i$  and  $\bar{\kappa}_i$  are mean of the (intact mode shape, subtractable from arrays of the vector  $\zeta_i$ ) dependent and mean of the (damaged mode shape, subtractable from arrays of the vector  $\kappa_i$ ) independent variables. Moreover,  $C_i$  are the quasi-correlation coefficients.  $n$  is the number of sampling points of the signal. Finally, the Pearson-based correlation coefficients mode shape index are defined as follows:

$$R(i) = \kappa_i - \frac{(\kappa_i - \bar{\kappa}_i)(\zeta_i - \bar{\zeta}_i)}{\sqrt{(\kappa_i - \bar{\kappa}_i)^2(\zeta_i - \bar{\zeta}_i)}} \quad (11)$$

### 2.3. One-Dimensional Discrete Wavelet Transform

An important type of wavelet transform is one-dimensional discrete wavelet transform. The one-dimensional discrete wavelet transforms for the signal  $R(i)$  are defined as follows:

$$R(i) = \kappa_i - \frac{(\kappa_i - \bar{\kappa}_i)(\zeta_i - \bar{\zeta}_i)}{\sqrt{(\kappa_i - \bar{\kappa}_i)^2(\zeta_i - \bar{\zeta}_i)}} = A_j(i) + \sum_{j < J} D_j(i) \quad (12)$$

where  $A_j$  show approximation signals at level  $j$ ,  $D_j$  are detail signals at level  $j$ .

The approximation signals at level  $j$  are obtained as follows [21]:

$$A_j(i) = \sum_{k=-\infty}^{+\infty} cA_{j,k} \phi_{j,k}(i) \quad (13)$$

where  $cA_{j,k}$  are approximation coefficients at level  $j$ .  $\phi_{j,k}(i)$  indicate scaling functions at level  $j$ .

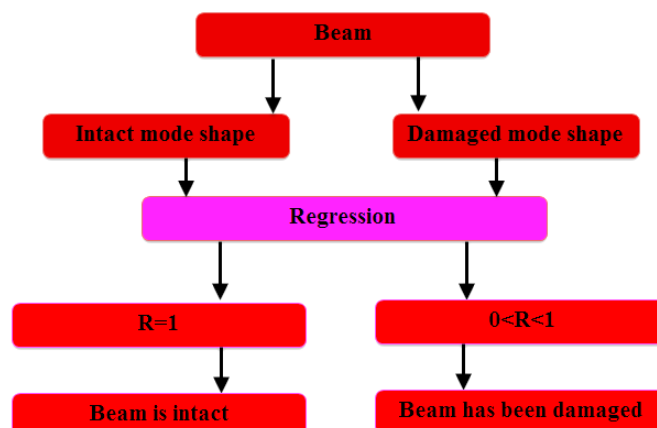
The detail signals at level  $j$  denote expressed as follows [21]:

$$D_j(i) = \sum_{k \in \mathbb{Z}} cD_{j,k} \psi_{j,k}(i) \quad (14)$$

where  $cD_{j,k}$  are detail coefficients at level  $j$ .  $\psi_{j,k}(i)$  indicate wavelet functions.

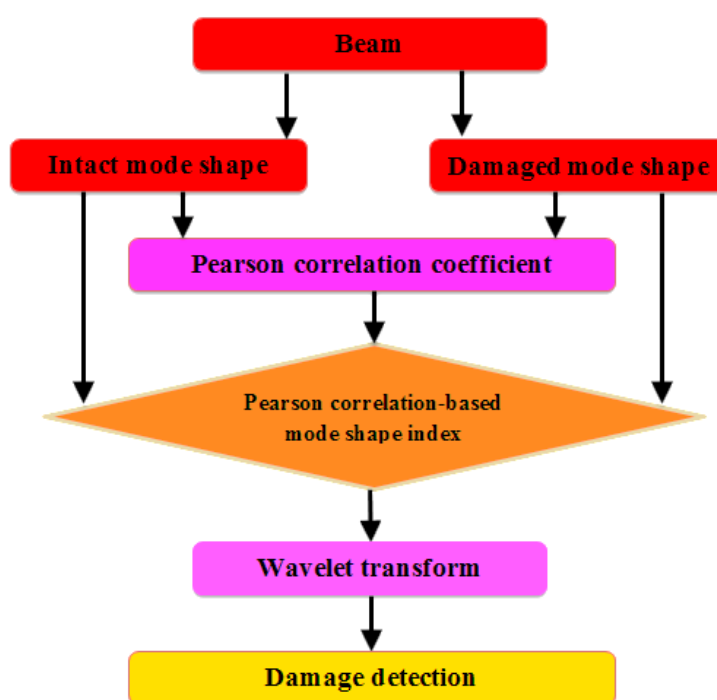
### 3. Proposed Approach

In this section, the proposed approach is presented. The approach is based on statistical features of intact and damaged mode shapes related to the beam structures, and discrete wavelet transforms. As seen in Figure 1, in the first step, the health of beam structure is monitored by the regression applied between intact and damaged mode shapes. For practical applications, the first intact and mode shapes can be used. Note that the pair mode shapes are obtained either experimentally or numerically. As seen in Figure 1, when the regression factor ( $R$ ) is equal to 1, it means that the beam structure is intact, and when it is between 0 and 1, it means that the beam structure has experienced damage, at least. If  $0 < R < 1$ , the damage detection process begins according to the flowchart shown in Figure 2.



**Figure 1.** Flowchart of beam structure's health monitoring by regression indicator.

As seen in Figure 2, for the proposed damage detection approach, the Pearson-based correlation coefficients between the intact and damaged mode shapes are calculated to use as the processing signal in the one-dimensional discrete wavelet transform instead of using a damaged mode shape signal. Therefore, in this study, damage detection is performed based on the Pearson-based correlation coefficients between the intact and damaged mode shapes in order to improve the power of damage detection by the one-dimensional discrete wavelet transform and eliminate the weakness related to selecting the best mother wavelet and vanishing moments.

**Figure 2.** Flowchart of beam structure's damage detection by combining the abilities of the Pearson-based correlation coefficient and the one-dimensional discrete wavelet transform.

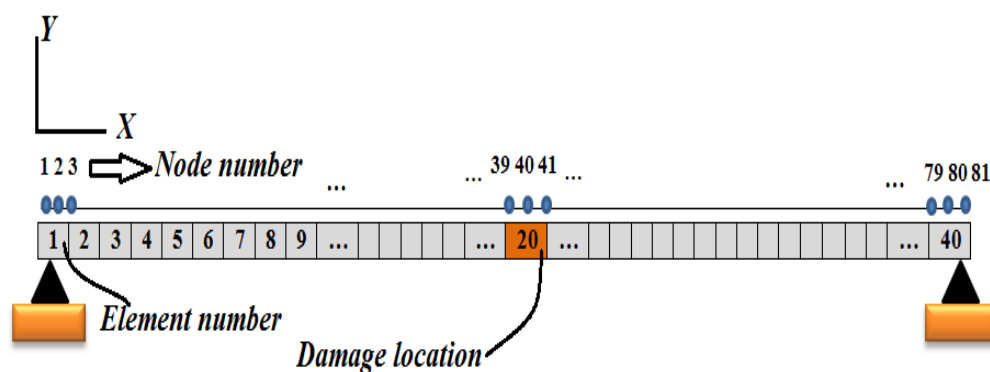
## 4. Results

### 4.1. Numerical Results

The FEM is applied to calculate the intact and damaged mode shapes of beam structure for the numerical investigation. As indicated in Figure 3, the beam is divided into 40 elements (81 nodes). The constant properties of the studied steel beam structure are tabulated in Table 1.

**Table 1.** The constant properties of studied beam structure.

Property	Symbol	Value (Unit)
Cross-sectional area (m <sup>2</sup> )	$A$	$1.82 \times 10^{-4}$
Moment of inertia (m <sup>4</sup> )	$I$	$1.46 \times 10^{-9}$
Density (kg/m <sup>3</sup> )	$\rho$	2685
Total length of beams (m)	$L$	0.5
length of each element (m)	$l_e$	0.1



**Figure 3.** The considered beam in numerical investigation.

According to the Figure 3, the damage is applied to element number twenty. Moreover, each element has three nodes. According to Table 2, in the numerical investigations, four damage scenarios are applied on the beam to investigate the performance of our proposed method. Damages are applied by reducing the stiffness of the twentieth element as following:

$$[K^e]_d = \alpha [K^e]$$

$$D = \frac{[K^e] - [K^e]_d}{[K^e]} \times 100 = \frac{[K^e] - \alpha [K^e]}{[K^e]} \times 100 = (1 - \alpha) \times 100 \quad (15)$$

where  $D$  is damage level,  $[K^e]_d$  is stiffness of the damage element. Moreover,  $\alpha$  is a constant ( $0 < \alpha < 1$ ).

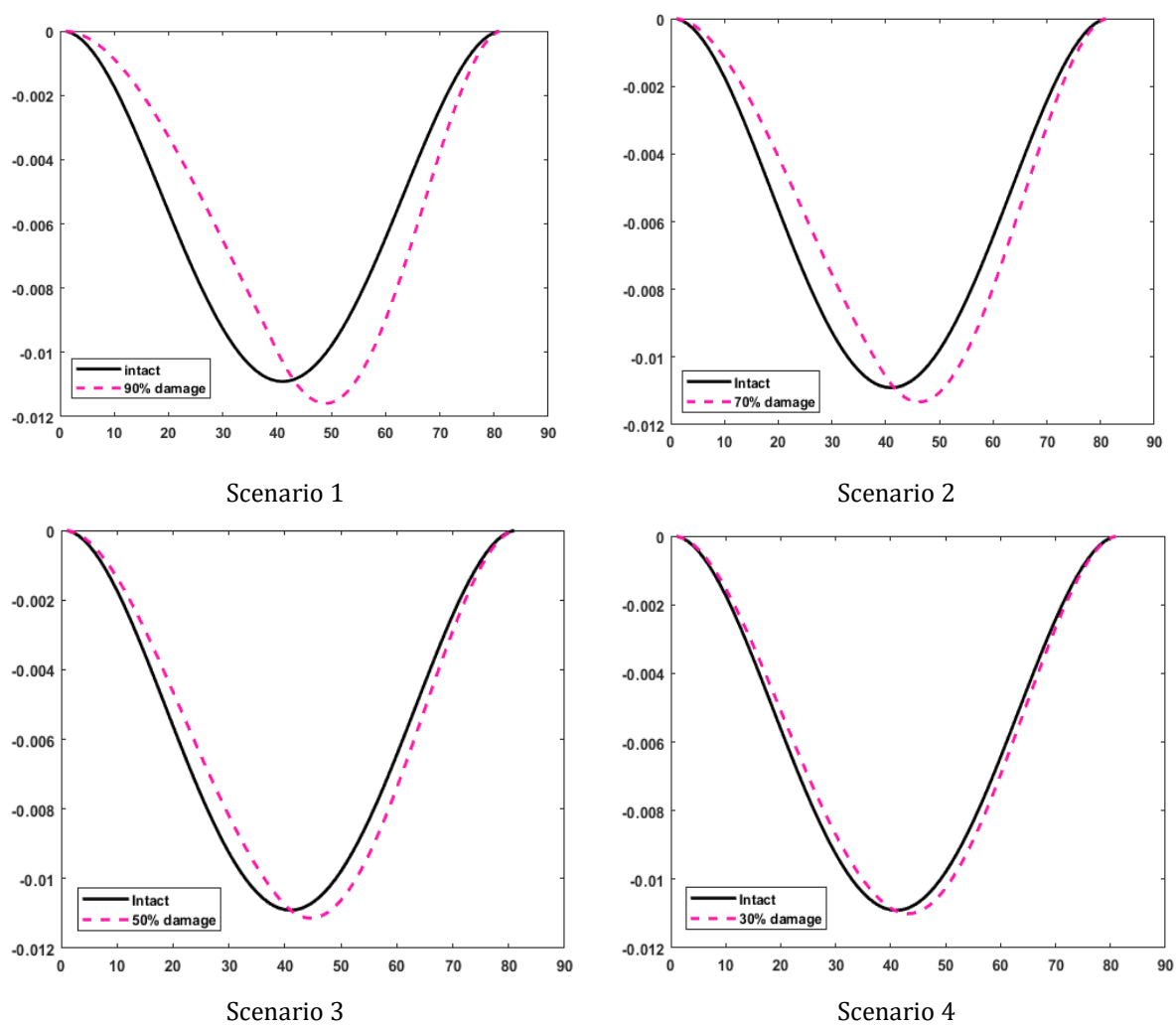
#### 4.2. Numerical Investigations for Structural Health Monitoring

In this section, numerical investigations are performed for structural health monitoring. According to Table 2, four different damage scenarios are considered to evaluate the performance of our regression-based approach for structural health monitoring of beam-like structures.

**Table 2.** Four damage scenarios for structural health monitoring.

Scenarios No.	Damage Location		Damage Level
	Element No.	Nodes No.	
1	20	39, 40, 41	90%
2	20	39, 40, 41	70%
3	20	39, 40, 41	50%
4	20	39, 40, 41	30%

The 90% damage is applied to the twentieth element (i.e., in nodal coordinate, nodes 39, 40, and 41) in the first damage scenario. In the second, third, and fourth damage scenarios, 70%, 50%, and 30% damages are applied to the same location (i.e., nodes 39, 40, and 41), respectively. Figure 4 shows eight mode shapes obtained from four different damage scenarios. As seen in this figure, damage causes the shift of the intact mode shapes in all scenarios. Moreover, results show that as the damage level increases, the shift increases. This finding was reported in the literature that stated that damage shifts mode shapes. As seen in Figure 4, the location of damage cannot be determined using mode shape directly. The damaged mode shapes provide a qualitative tool to structural health monitoring. This qualitative way is considered a weakness when level of damage is low. For fixing this problem, this paper proposes a quantitative approach called regression.



**Figure 4.** Differences in intact and damaged modes for four considered damage scenarios.

As mentioned, the regression index  $R$  can show health of beam, if the regression index is  $R = 1$ , then the beam structure is intact, and when it is between 0 and 1, it means that the structure has been experienced the damage, at least. Figure 5 shows the regression diagram between the first mode and the first mode. As expected, the value of the regression index is  $R = 1$ .

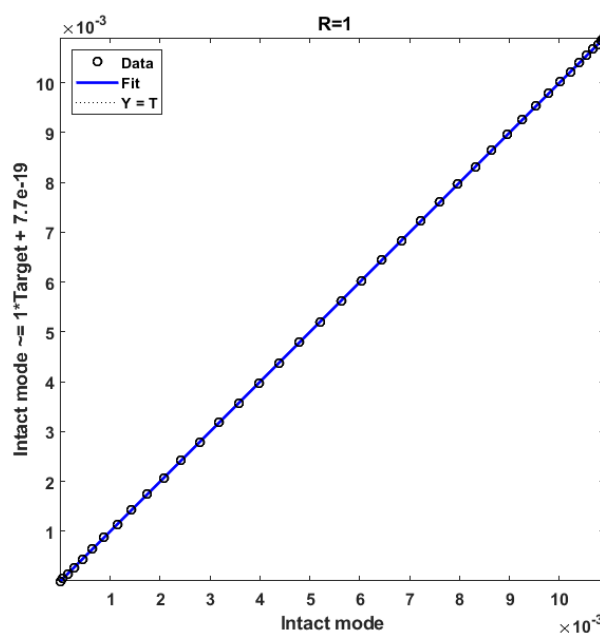
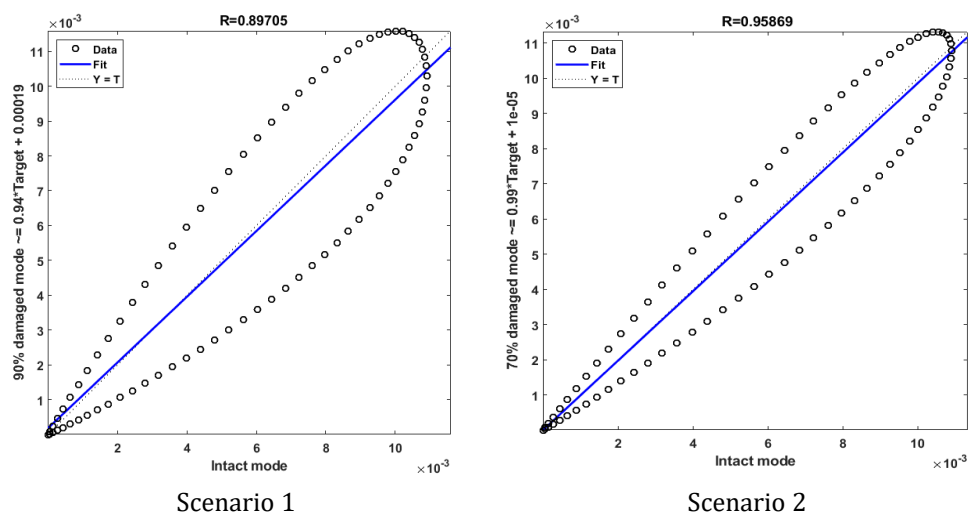


Figure 5. Regression diagram for two intact modes.

On the other hand, Figure 6 shows the regression diagram for the four considered scenarios. This figure shows that the regression index  $0 < R < 1$  is different for the four different scenarios considered. The findings demonstrate that as the damage level increases, the  $R$  decreases. Moreover, Figure 6 shows that even for the low-level damages, reporting  $R$  is possible. In contrast, when the level of damage, it is challenging to detect the difference between intact and damaged modes.

After it is specified using the regression index that the beam is damaged. Our proposed method is tested. Figure 7a shows the original signal  $s$  (damaged mode shape) and the results of their one-dimensional discrete wavelet transform.

After it is specified using the regression index ( $R$ ) that the beam is damaged. Our proposed method is tested. Figure 7a shows the original signal  $s$  (damaged mode shape) and the results of its one-dimensional discrete wavelet transform. The wavelet family Symlet decomposes the signal  $s$ , and the vanishing moment is equal to 2 for this figure. The damage level in this figure equals 90%, and the damage is located at the twentieth element.



Scenario 1

Scenario 2



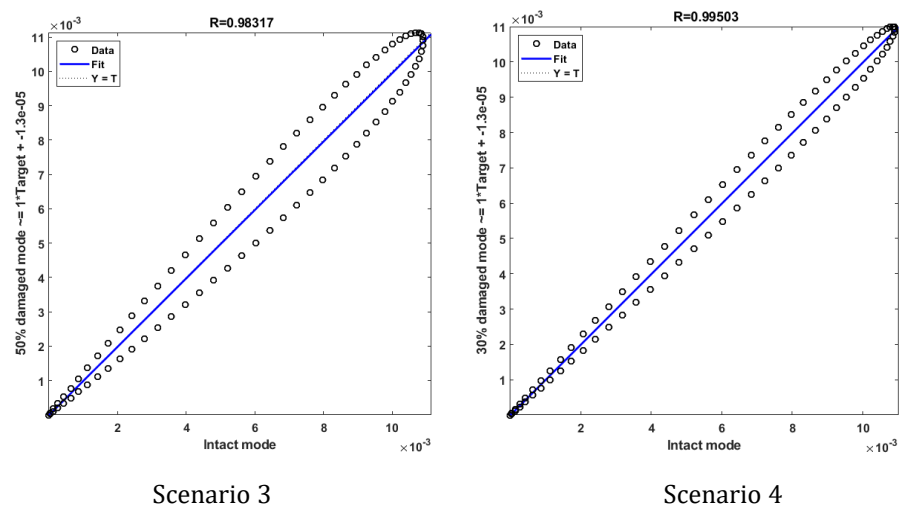


Figure 6. Regression diagram for the four considered scenarios.

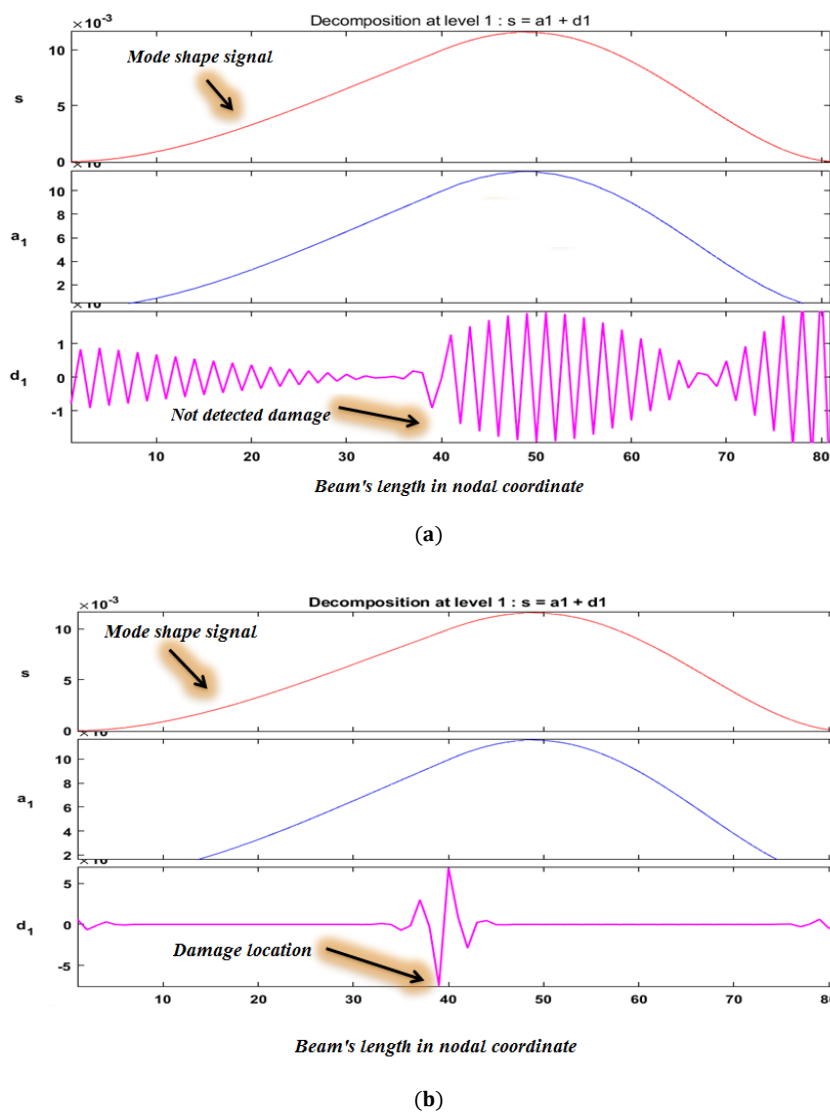


Figure 7. The effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from damaged mode shape signal by 90% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.

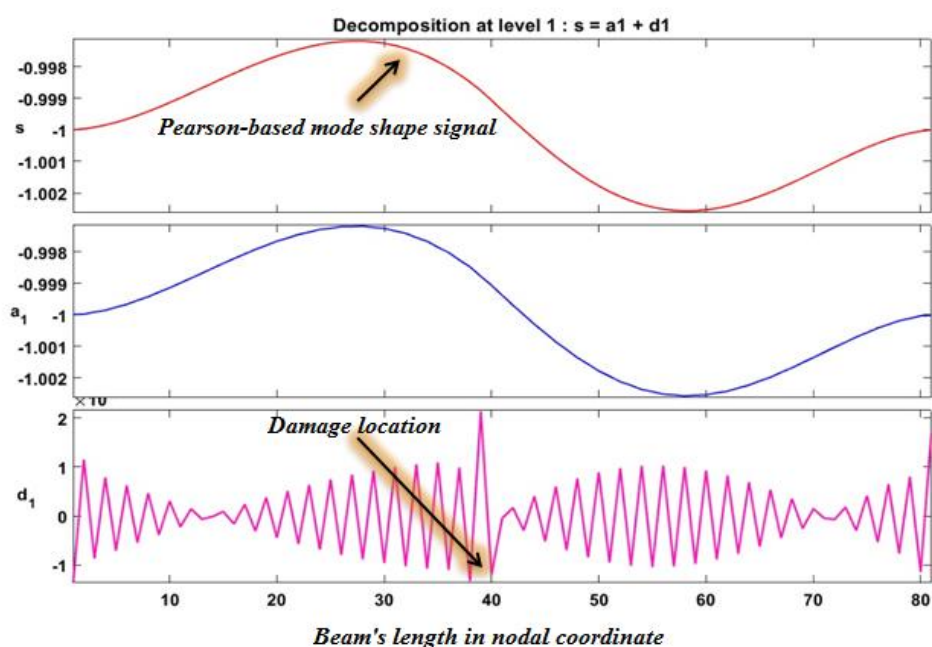
### 4.3. Numerical Investigations for Damage Detection

After the structural health monitoring health phase, if  $0 < R < 1$ , the beam structure has experienced damage. Thus, damage detection should be performed to prevent further losses. This section presents numerical investigations for damage detection. According to Table 3, three different damage scenarios are considered for evaluating the performance of our proposed damage detection method. Similar to the structural health monitoring health phase, in this section, damages of all damage scenarios are located at element number 20 (i.e., in nodal coordinate, nodes 39, 40, and 41). In the first, second, and third scenarios, damages levels are 90%, 70%, and 10%.

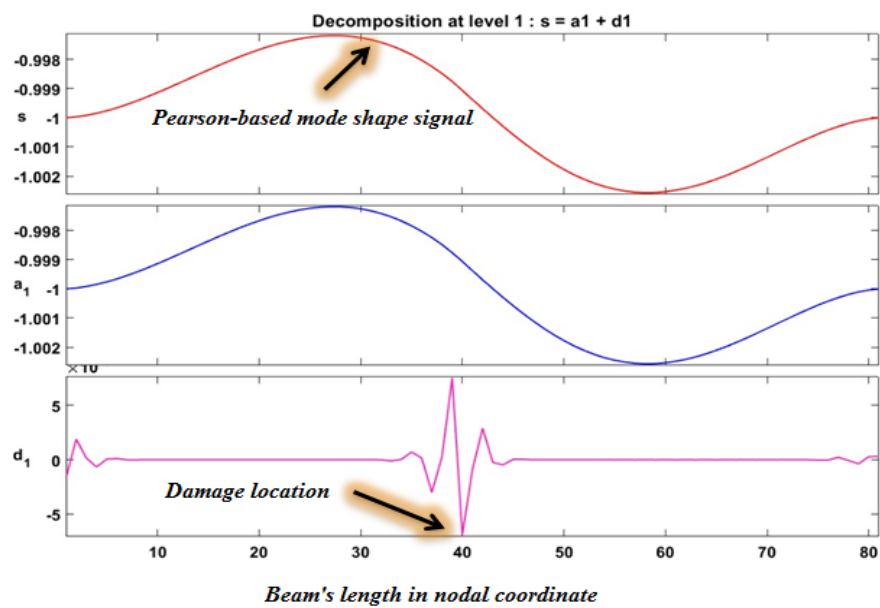
**Table 3.** Three damage scenarios for structural health monitoring.

Scenarios No.	Damage Location		Damage Level
	Element No.	Nodes No.	
1	20	39, 40, 41	90%
2	20	39, 40, 41	70%
3	20	39, 40, 41	10%

Results of numerical investigations for damage detection are presented in Figures 7–13. These results prove that the performance approach acts better than the conventional. Figures 7b, 8b, 9b, 10b, 11b, 12b and 13b demonstrate that using the vanishing moments 5, damage detection both by model shape signals and Pearson-based mode shape signal brings accurate results. However, for vanishing moments 2, the proposed Pearson-based mode shape signals bring better damage detection results than the conventional mode shape signals.

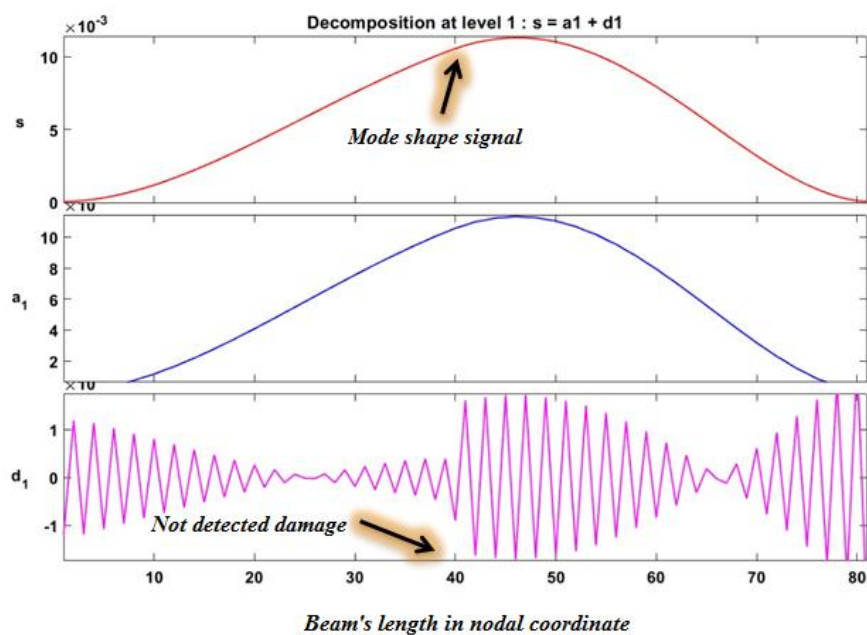


(a)

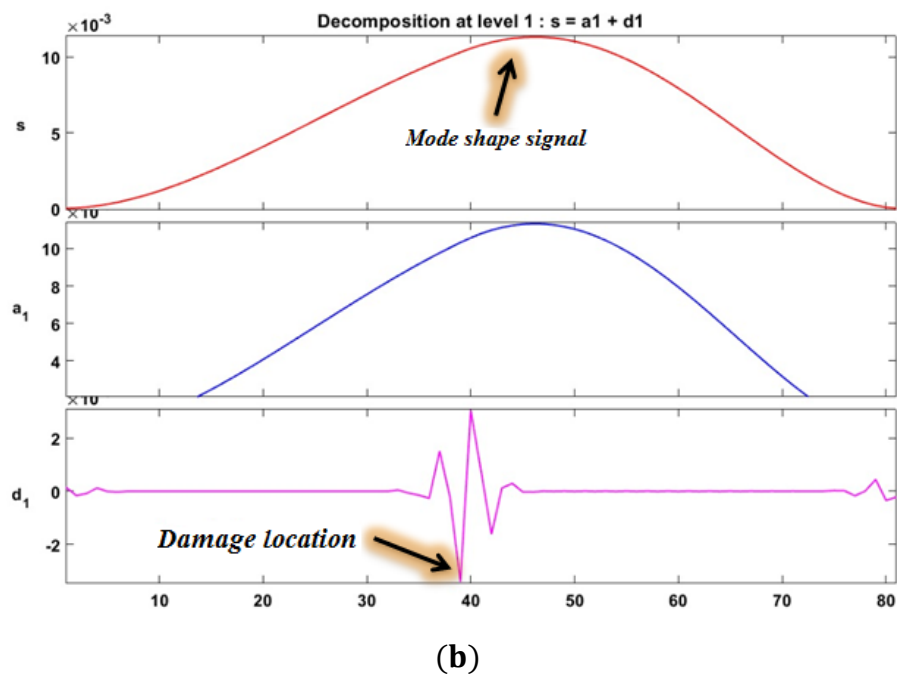


(b)

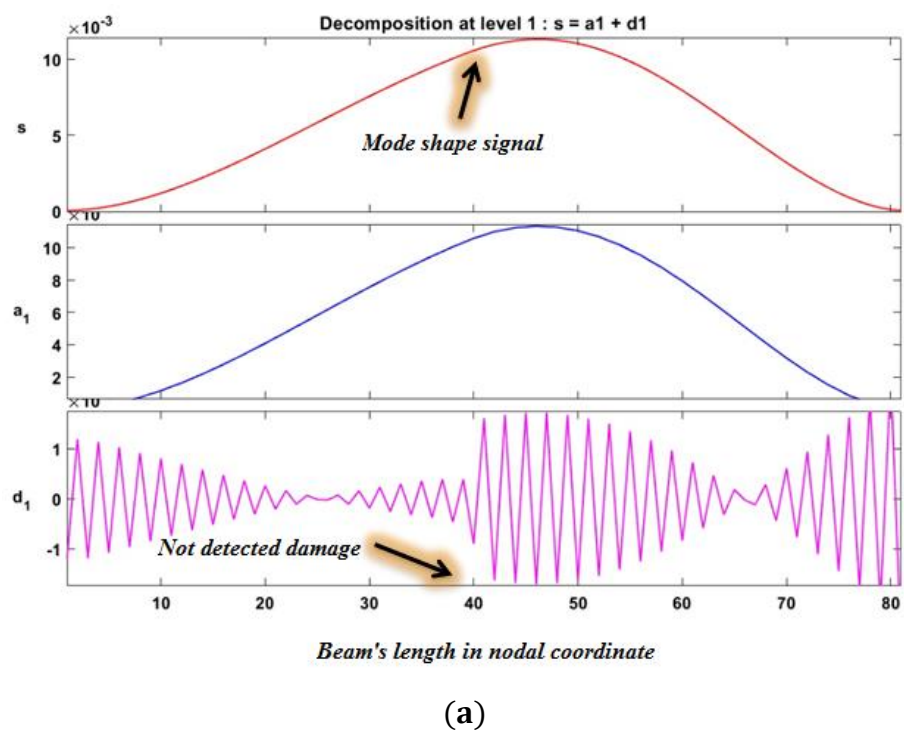
**Figure 8.** The effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from Pearson-based mode shape signal by 90% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.

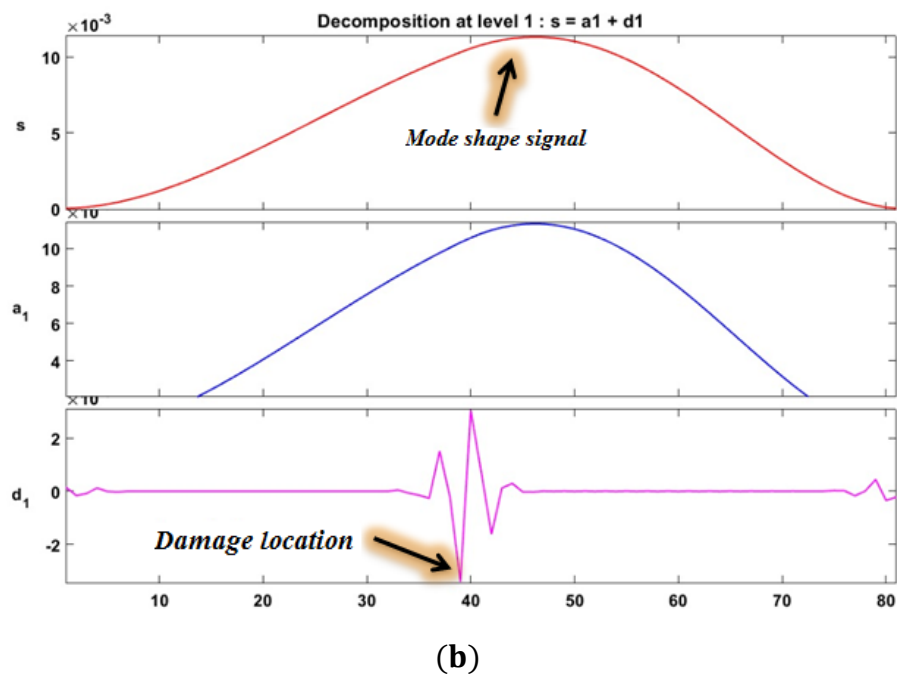


(a)

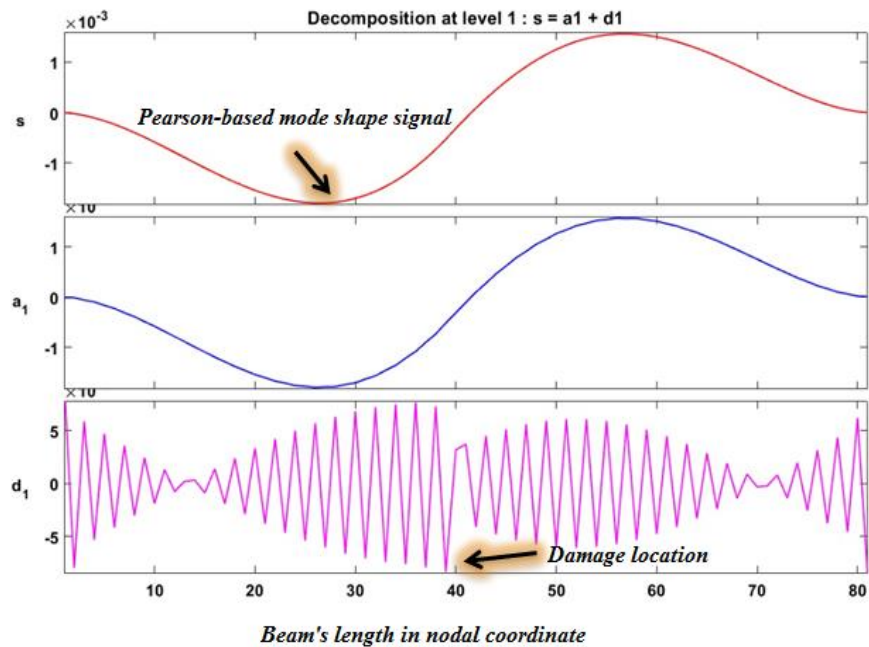


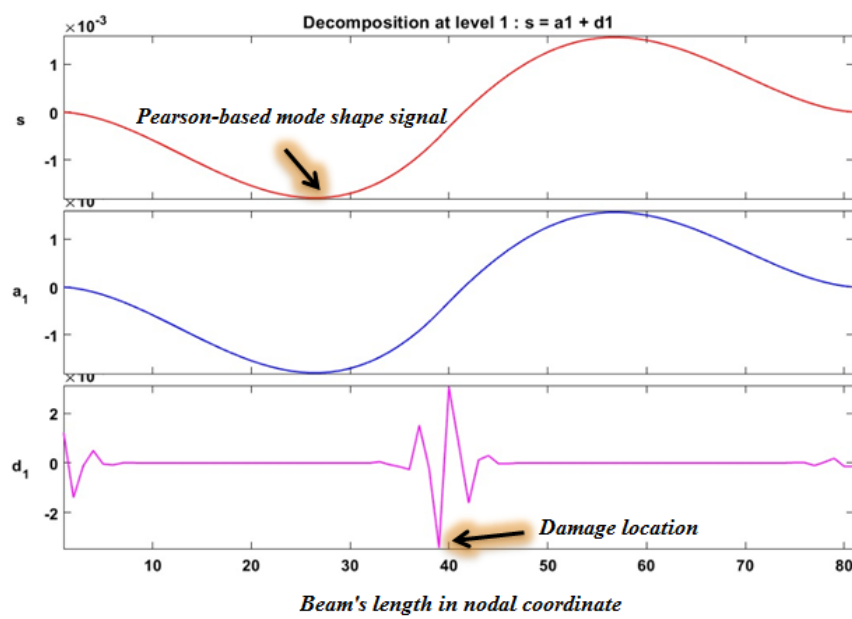
**Figure 9.** The effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from damaged mode shape by 70% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.





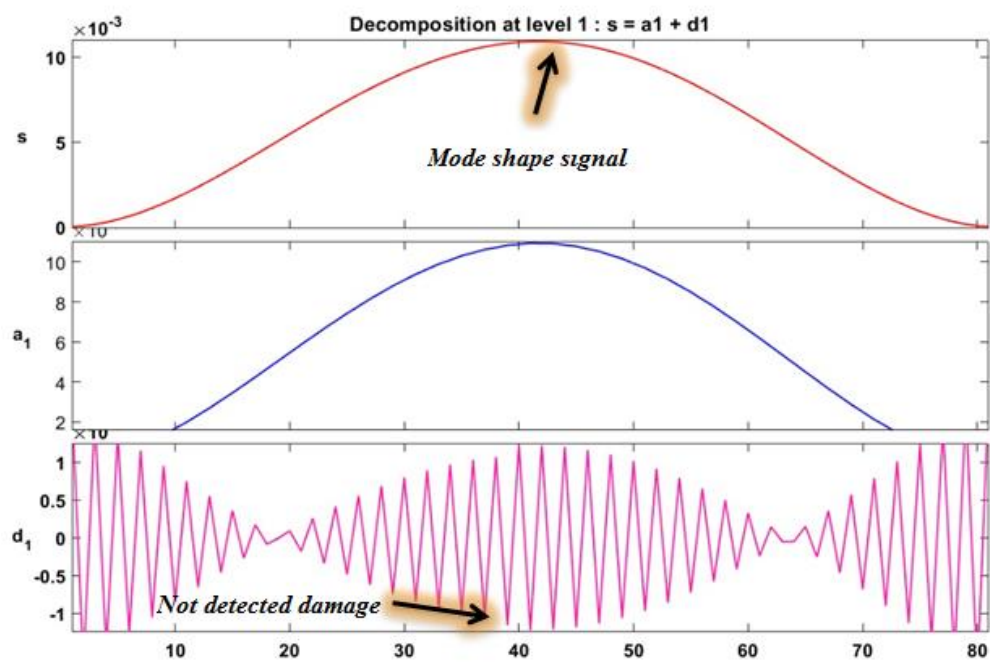
**Figure 10.** The effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from damaged mode shape by 70% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.



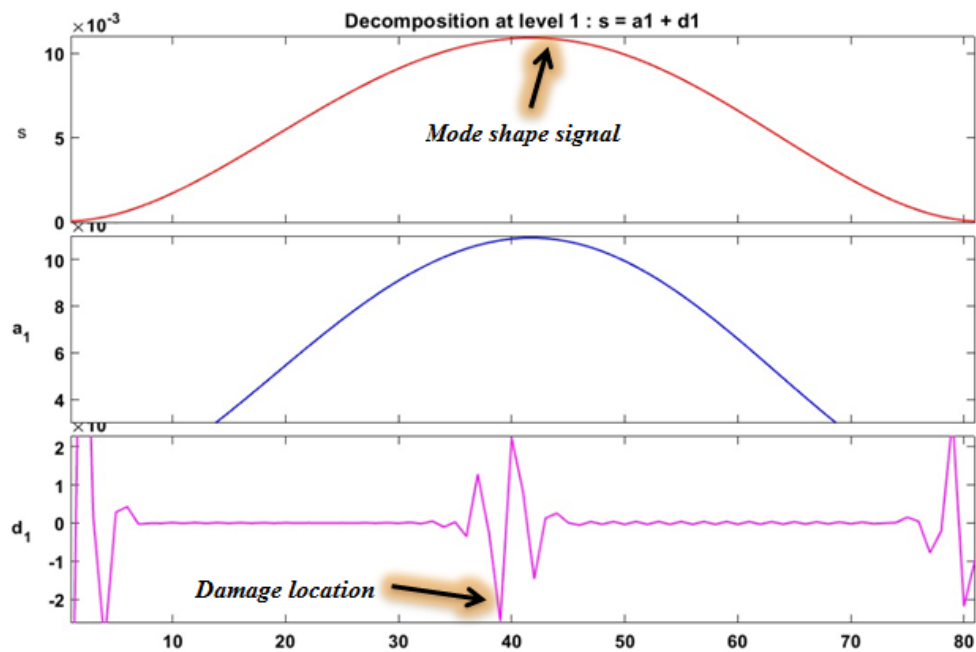


(b)

**Figure 11.** The effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from the Pearson-based coefficient extracted from intact and damaged mode shapes by 70% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.

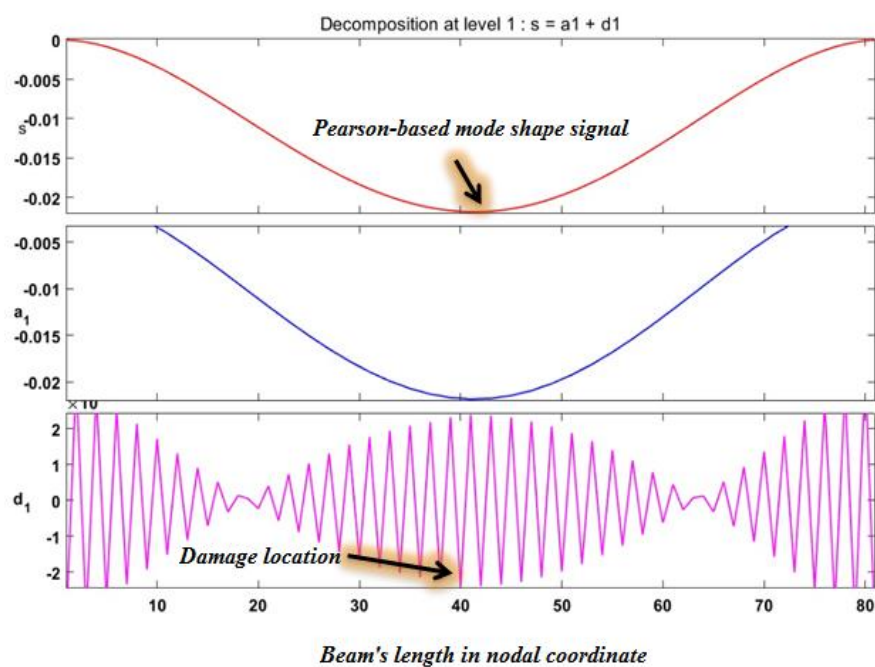


(a)



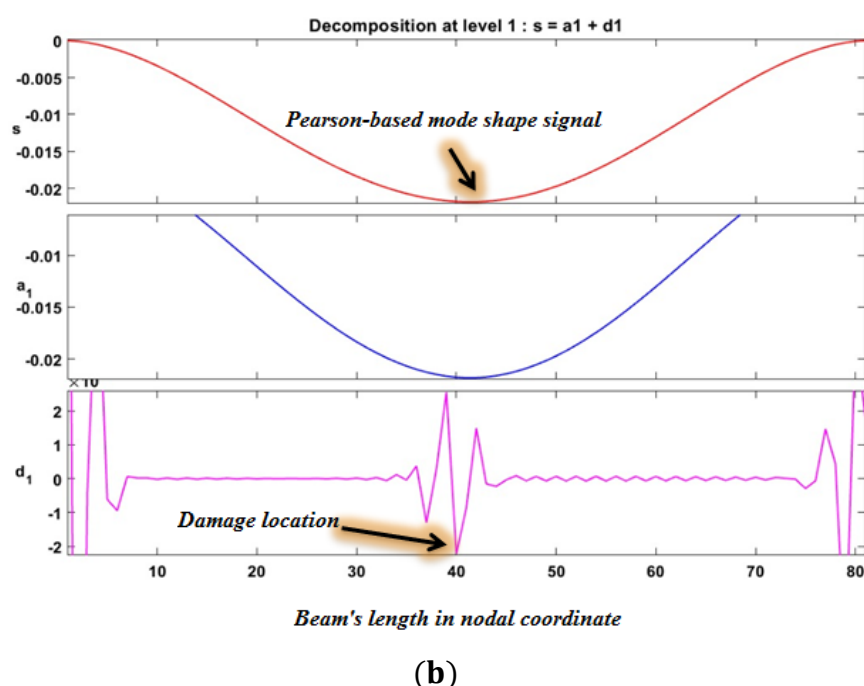
(b)

**Figure 12.** the effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from the Pearson-based coefficient extracted from intact and damaged mode shapes by 10% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.



(a)





**Figure 13.** The effect of increasing the vanishing moments on the accuracy of damage detection by 1D-DWT from the Pearson-based coefficient extracted from intact and damaged mode shapes by 10% damage: (a) vanishing moments = 2, (b) vanishing moments = 5.

The results of the one-dimensional discrete wavelet transform for mode shape signals and Pearson-based signals for two vanishing moments (i.e., 2 and 5) are shown in Figures 7–13 to simultaneously investigate the effect of both using the proposed Pearson-based signals in the result and increasing the order of vanishing moments. In the results presented in Figures 7–13,  $a_1$  indicates the approximation signal at level 1, and  $d_1$  indicates the detail signal at level 1. The approximation  $a_1$  is the high-scale, low-frequency component of the original signal. The detail  $d_1$  is the low-scale, high-frequency component of the original signal. Therefore, it is expected to search the damage's location in the detail signals.

By comparing the images shown in Figures 7–13, it is found that our proposed approach eliminates the need for selecting vanishing moments. It is also seen that our proposed method introduces a good alternative for mode shape signals to improve the accuracy of damage detection. Thus, using the Pearson-based correlation coefficients of the mode shapes instead of the mode shapes in the wavelet transform can improve the damage detection's accuracy, especially in low-level damages. Moreover, the findings show that the damage location, if detected, appears in the detail signals and the approximation signals, as expected, are a general approximation of the signals. The results show that it is possible to identify damage and cracks in the structure using wavelet transformation of Pearson-based correlation coefficients at level 1; as a result, there is no need to choose vanishing moments. This is an important advantage over the conventional wavelet transform because it eliminates trial and error efforts for the selection of the best mother wavelet function.

#### 4.4. Experimental Results

Careful design and maintenance of various structures such as buildings, bridges, dams, airplanes, trains, beams, etc., is essential. One of the requirements for the design and maintenance of structures is their dynamic analysis. Due to the unavailability of analytical solutions for complex structures, with different loads and boundary conditions, as well as errors such as errors resulting from the application of inappropriate assumptions



and theories, errors in modeling the details of complex structures, and lack of accurate information about the properties materials, numerical approximation models such as the finite element method also face problems. Therefore, modal testing is a suitable tool to achieve the dynamic properties of the structure.

In order to implement the modal test, the intact steel beam is used, as shown in Figure 14. Then, according to Figure 14a, the intact beam is cracked to be used to have the corresponding cracked beam, as shown in Figure 14. The beams are suspended with a soft clamp to implement free-free boundary conditions as seen in Figure 14b. A DJB A120V accelerometer is used to measure acceleration. This accelerometer is installed at point 4 of the structure. An 8202BK vibrating hammer equipped with an 8200BK dynamometer is used to apply force to the structure. A 2647A amplifier is also used to convert the power signal. The steel beam is divided into 10 points. The accelerometer is fixed at point 4, and at all points, the impact force is applied with a hammer. This test is performed once for a healthy beam and once for a damaged beam.



**Figure 14.** Intact steel beam and its divisions.

In order to have damage on the steel beam, at a distance of 30 cm from the head of the beam, slot-shape damage with 0.5 mm deep and 1 mm wide across the width of the beam is created using a CNC milling machine (Figure 15a). Also, Figure 15b shows the cracked steel beam and its free-free boundary conditions.



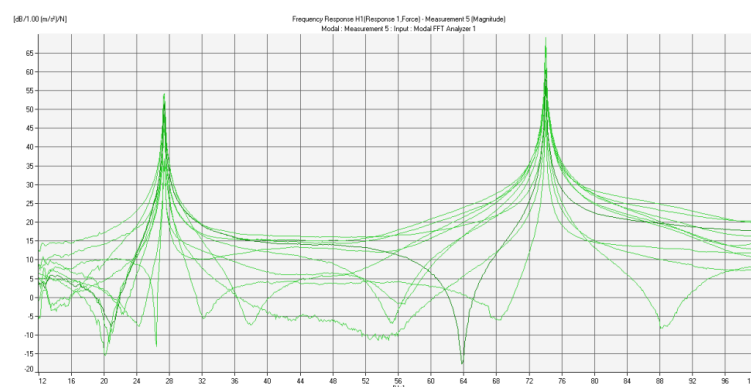
(a)



(b)

**Figure 15.** (a) Using CNC milling machining for creating damage on the steel beam, (b) the cracked steel beam and its free-free boundary conditions.

As mentioned, the modal test is performed at all beams' points (the accelerometer is fixed at point 4, and the hammer is moved at other points). The test frequency range is 0 to 100 Hz. The frequency response functions obtained for the intact beam (Figure 16) have been calculated using the BK3560D analyzer and Pulse 8 software by performing experiments and measuring force and acceleration signals.



**Figure 16.** The frequency response functions obtained for the intact beam.

At the end of the experiment, the mentioned frequency response functions are sent to the Icats software and analyzed. After this analysis, the natural frequencies, damping coefficients, and the beam modes' shape are obtained. The natural frequencies and damping coefficients are reported in Table 4. Moreover, the normalized mode shapes of the steel beam in the intact conditions is presented in Table 5.

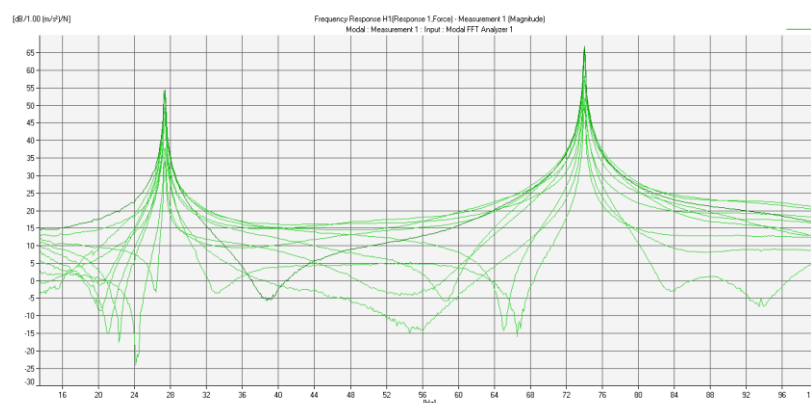
**Table 4.** Experimental natural frequencies and damping coefficients of the intact steel beams.

Mode Number	1	2
Frequency (Hz)	27.309	73.9782
Structural Damping coefficient	0.0037	0.0019

**Table 5.** Experimental normalized mode shapes obtained from the modal test of the intact steel beam.

Node Number	Mode 1	Mode 2
1	−2.599334	−2.051736
2	−1.076546	0.222916
3	0.260243	1.82904
4	1.125964	1.509068
5	1.677929	0.790353
6	1.68134	−0.780846
7	1.246111	−1.89759
8	0.294394	−2.135082
9	−0.808516	−0.486373
10	−2.331891	1.687432

Then, after applying damage to the beam, the test is performed again in the same way as before. The accelerometer is installed at point 4, and a hammer blow is applied at all points. The test frequency range is 0 to 100 Hz. The frequency response functions obtained for the damaged beam (Figure 17) have been calculated using the BK3560D analyzer and Pulse 8 software by performing experiments and measuring force and acceleration signals.



**Figure 17.** The frequency response functions obtained for the damaged beam.

After testing, the obtained frequency response functions are sent to the Icats software and analyzed. After this analysis, the natural frequencies, damping coefficients, and the damaged beam modes' shape are obtained. The natural frequencies and damping coefficients are reported in Table 6. Moreover, the normalized mode shapes of the steel beam in the damaged conditions are presented in Table 7. The plot of experimental intact and cracked mode shapes is presented in Figure 18 with the red line and blue line, respectively.

Experimental findings demonstrate that damage causes to reduce natural frequencies (Tables 3 and 5). Moreover, similar to the numerical study, damage causes shift mode shapes (Figure 18). Experimental results verify that our proposed method for structural health monitoring is effective (Figure 19). In addition, the experimental findings verify that our proposed method improves the performance of the conventional 1D-DWT (Figures 20 and 21).

**Table 6.** Experimental natural frequencies and damping coefficients of the damaged steel beams.

Mode Number	1	2
Frequency (Hz)	27.303	73.9384
Structural Damping coefficient	0.0035	0.0012

**Table 7.** Experimental normalized mode shapes obtained from the modal test of the damaged steel beam.

Node Number	Mode 1	Mode 2
1	−2.455124	−1.785898
2	−1.153495	0.273122
3	0.209485	1.536691
4	1.168128	1.984244
5	2.038565	0.724189
6	1.864971	−0.814256
7	1.316461	−1.883213
8	0.365667	−1.860149
9	−0.942927	−0.424004
10	−2.225317	1.283471

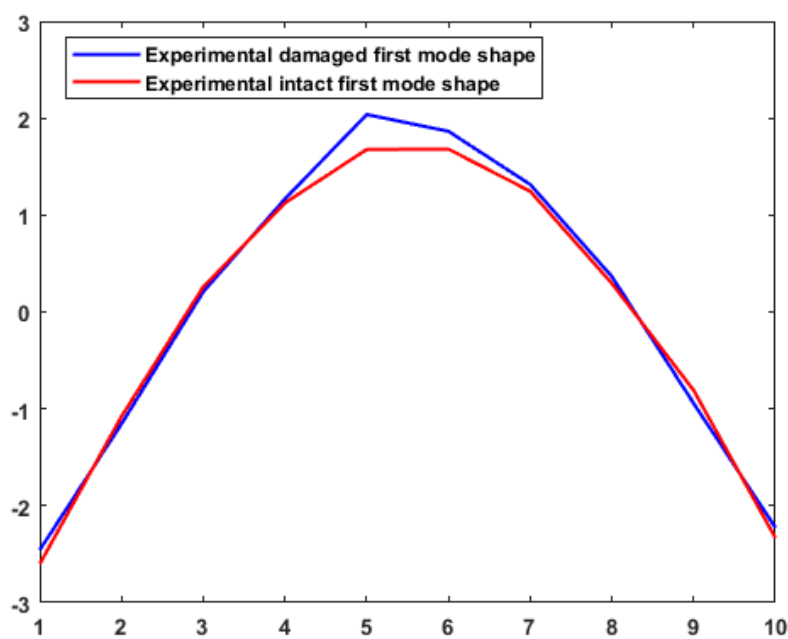


Figure 18. Experimental damaged and intact first mode shapes.

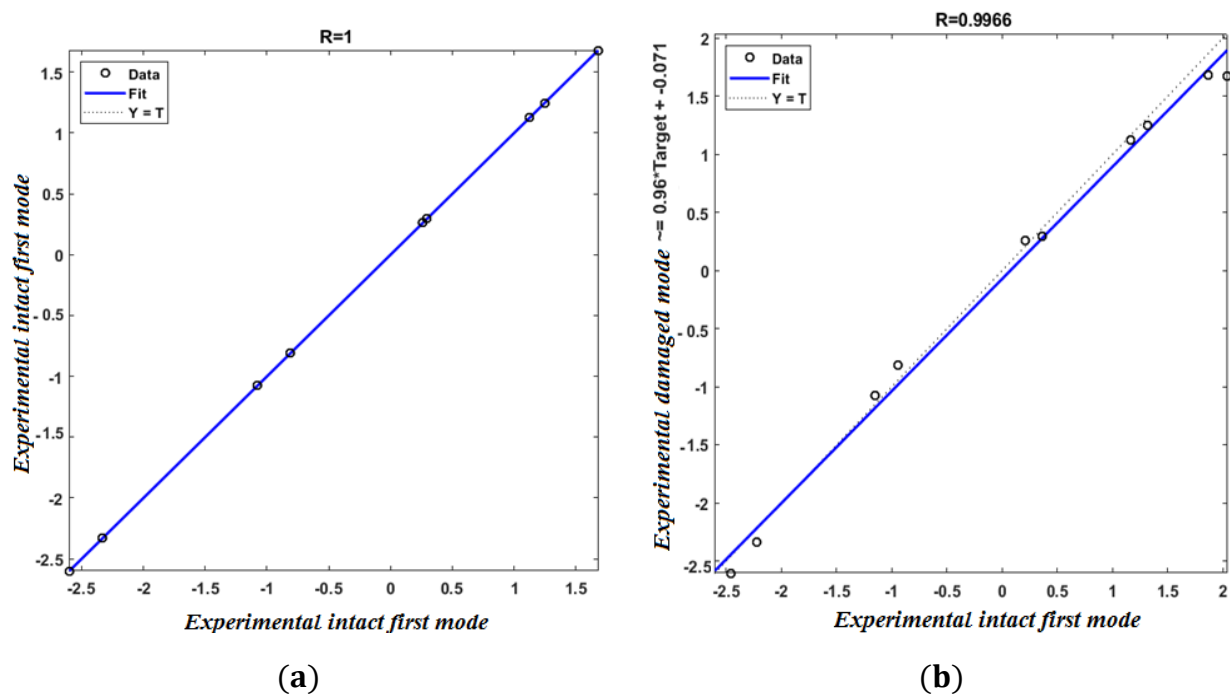


Figure 19. Regression plot for (a) regression diagram for two experimental intact mode shapes and (b) regression diagram for experimental damaged and intact mode shapes.

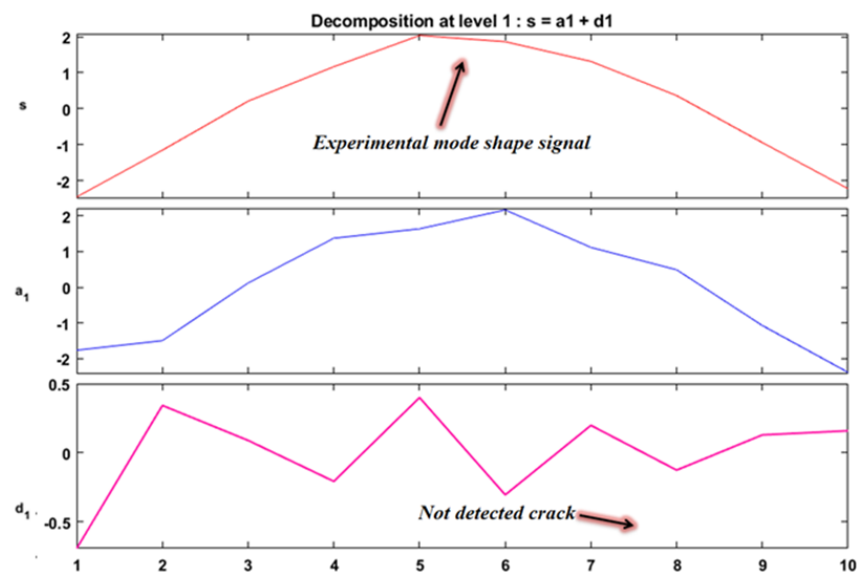


Figure 20. Damage detection by 1D-DWT from experimental damaged mode shape.

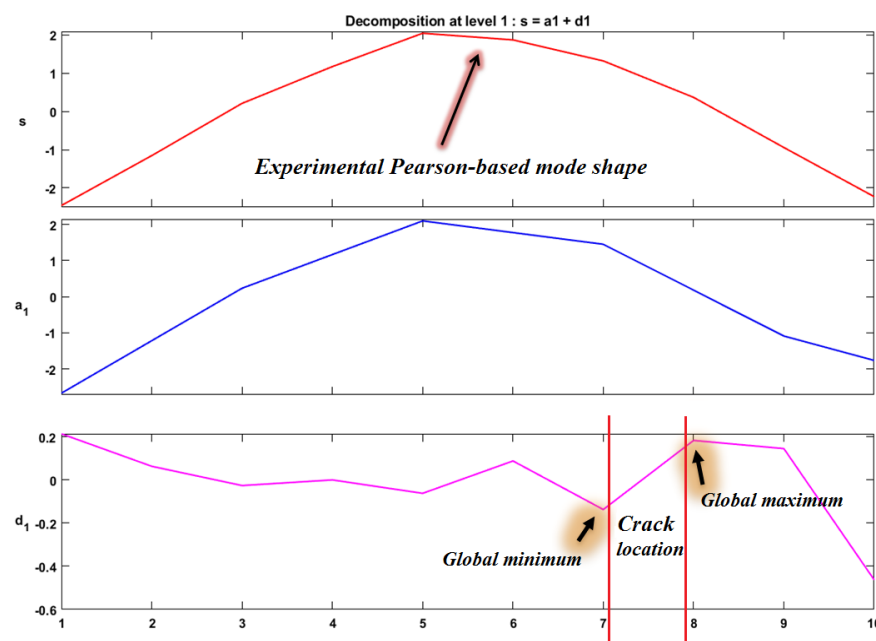


Figure 21. Damage detection by 1D-DWT from the Pearson-based coefficient extracted from experimental intact and damaged mode shapes.

## 5. Conclusions

In this study, a novel strategy for structural health monitoring and damage detection of the beam structure is present to improve the efficiency of the one-dimensional discrete wavelet transforms 1D-DWT. For structural health monitoring of beam structures, it is proposed to use regression index (R) to know damage conditions numerically. The regression index is beneficial, especially for level damages and great damage-sensitive application of beam structures. There is no apparent difference between intact and damaged mode shapes under low-level damages. Findings show that the performance approach acts better than the conventional. In addition, findings demonstrate that using the vanishing moments 5, damage detection both by model shape signals and Pearson-based mode shape signal brings accurate results. However, for vanishing moments 2, the proposed

Pearson-based mode shape signals bring better damage detection results than the conventional mode shape signals. In order to verify the performance of our proposed method in practice, experimental investigations are performed. The experimental results demonstrate the efficiency of using the Pearson-based coefficients of vibrational mode shapes of the beam structures instead of damaged mode shaped to process by 1D-DWT.

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